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Theoretical study on impingement heat transfer with single-phase free-surface slot jets

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Abstract

A theoretical analysis was conducted to characterize heat transfers from horizontal surfaces to normally impinging slot jets under arbitrary-heat-flux condition. The thermal and hydraulic boundary layers of laminar flow were divided into four regions of flow along heat transfer surfaces including a stagnation zone and three wall jet zones, from which general expressions of heat transfer coefficients were obtained. Furthermore, these results were compared with experimental and analytical data available in published literature. Good agreements were observed from the comparisons. © 2005 Elsevier Ltd. All rights reserved.

Keywords: Slot jet impingement; Heat transfer; Laminar boundary layer; Theoretical analysis

1. Introduction

Impinging jets have been widely used in industrial cooling or heating processes as a means of providing very high heat and mass transfer rates. In most cases air is employed as a working medium in jet impingement and a large amount of research on impinging air jets has been reported [1,2]. Recently, more attention has been directed to the study of liquid jet impingement as the heat transfer rates can be increased several orders of magnitude in comparison with that of gas jets. Research on liquid jets was stimulated by their possible application to the cooling of heat engines [3,4] or electronic de-

vices [5–7], as well as in the thermal treatment of metals [8,9].

Two principal jet configurations are relevant to the application of jet impingements: circular jets and slot jets, but for which flow and heat transfer mechanics are distinctly different. A significant amount of research has been published on the heat and mass transfer with circular impinging jets [1,2,10,11]. Recently, however studies of heat and mass transfer with slot jet impingement have attracted more attention from scientists and engineers. Slot jet impingement becomes remarkable as it offers more beneficial features, such as higher cooling effectiveness, greater uniformity and more controllability. All these factors are suitable to countering the trend of continuously increasing heat flux and decreasing dimensions in compact electronics packages [12–16].

Schwarz [14] presented measurements and an analysis on the flow characteristics of two-dimensional jet

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Nomenclature			
В	slot nozzle width	$x_{\rm t}$	lateral location where thermal boundary
C_p	specific heat		layer reaches free surface
\hat{H}	liquid film thickness	$\chi_{ m v}$	lateral location where viscous boundary
h	heat transfer coefficient		layer reaches free surface
k	thermal conductivity	y	coordinate normal to impingement plate
L	sectional length of slot nozzle	Z	nozzle-to-plate spacing
Nu	=hB/k, Nusselt number		
Pr	$=\mu C_p/k$, Prandtl number		symbols
Q	jet flow rate	δ	hydrodynamic boundary layer thickness
q	wall heat flux	Δ	thermal boundary layer thickness
Re	$=V_0B/v$, Reynolds number	Δ_1	see Eq. (21)
T	local temperature	3	$= \int_0^x q \mathrm{d}x/qx$
$T_{ m w}$	local wall temperature	η	$=y/\delta$
T_0	jet temperature at nozzle exit	θ	dimensionless temperature
T_{∞}	temperature outside thermal boundary layer	μ	dynamic viscosity
U_{∞}	velocity outside flow boundary layer	v	kinematic viscosity
u	lateral velocity component	ξ	$=\Delta/\delta$
V_0	jet velocity at nozzle exit	ho	fluid density
v	vertical velocity component	Λ	$=\left(\frac{\delta}{B}\right)^{2}Re$
X	coordinate parallel to impingement plate	α	$=rac{Hq}{k(T_{ m w}-T_{\infty})}$

impingement, but only for the turbulent wall-jet zone. Kendoush [15] studied the heat and mass transfer mechanics of an impinging slot jet by means of boundary-layer theory for laminar flow. The results were restricted just to the stagnation zone. Park [16] and Chiriac [17] reported the numerical solutions of stream and heat transfers in stagnation and wall jet zones for laminar flow, respectively, whereas the solutions to the confined slot jets were available only for a limited range of Reynolds numbers. The objective of the present study is to provide detailed theoretical solutions on laminar flow for free-surface slot jet impinging onto horizontal surfaces under arbitrary-heat-flux conditions. The influence of Prandtl numbers on heat and mass transfers is also considered.

2. Theoretical analyses

The stream of slot jet impingement is shown in Fig. 1, in which the flow field along a heated surface is divided into four regions including a stagnation zone and three wall-jet zones based on the characteristics of flow and heat transfer for $Pr \geqslant 1$, respectively. In this paper predicative formulas are provided for the entire range of lateral flow under arbitrary-heat-flux conditions.

According to boundary layer theory [18], the flow in a liquid film layer satisfies the following conservation equations:

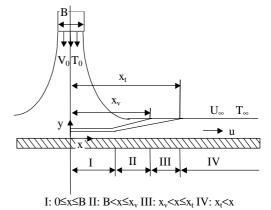


Fig. 1. Boundary layers of a free-surface slot jet impinging onto a horizontal plate.

$$\begin{cases} \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0\\ u\frac{\partial u}{\partial x} + v\frac{\partial u}{\partial y} = -\frac{1}{\rho}\frac{dP}{dx} + v\frac{\partial^2 u}{\partial y^2} \end{cases}$$
(1-2)

with the general conditions:

$$\begin{cases} y = 0, & u = v = 0 \\ y = \delta(y \to \infty), & u = U_{\infty}(x), & \frac{\partial u}{\partial y} = 0 \end{cases}$$
 (3-4)

$$2L \int_0^H u \, \mathrm{d}y = Q = V_0 \cdot (BL) \tag{5}$$

In region (I), stagnation zone, the speed outside the flow boundary layer is

$$U_{\infty} = V_0 \cdot (x/B) \tag{6}$$

So, the classic four-power polynomial can be used to approximate the velocity profile satisfying Eqs. (3–4):

$$\frac{u}{U_{\infty}} = f(\eta) = (2\eta - 2\eta^3 + \eta^4) + \frac{\Lambda}{6}\eta(1-\eta)^3 \tag{7}$$

where $\eta = \frac{y}{\delta}$, $\Lambda = \frac{\delta^2}{v} U_{\infty}' = \frac{\delta^2}{B^2} Re$. With the conditions that δ is finite at x = 0, δ and can be obtained by integrating Eq. (2):

$$\delta = 2.749BRe^{-0.5}, \quad \Lambda = 7.55$$
 (8)

Consequently, we obtain

$$f(\eta) = (2\eta - 2\eta^3 + \eta^4) + 1.260\eta(1 - \eta)^3 \tag{9}$$

In region(II), the speed U_{∞} remains almost constant, equal to V_0 , because the liquid here is unaffected by the viscous stresses. The velocity profile may be expressed by

$$\frac{u}{V_0} = \frac{3}{2}\eta - \frac{1}{2}\eta^3 \tag{10}$$

Substituting Eq. (10) into Eq. (2) yields

$$\frac{\mathrm{d}\delta^2}{\mathrm{d}x} = \frac{280}{13} \frac{v}{V_0} \tag{11}$$

Using the continuous condition of δ between regions (I) and (II) the value of δ in region II can be determined:

$$\delta \cong 2.9517B^{0.281}x^{0.719}Re^{-0.5} \tag{12}$$

Then, the hydrodynamic boundary layer of liquid film flow will be developed until right up to the free surface of stream where lateral distance x_v can be derived from Eqs. (10)–(12):

$$x_{\rm v} = 0.1627BRe^{0.695} \tag{13}$$

In regions (III) and (IV), as $x > x_v$, the viscous stresses become appreciable right up to the free surface, and the entire flow is of boundary layer type. We are able to use the velocity profile (10), but the free stream speed varies with x. The integral momentum equation should be revised, because in their equation the pressure gradient is zero:

$$\frac{\mathrm{d}\delta_2}{\mathrm{d}x} + \delta_2 \left(2 - \frac{\int_0^H \frac{u}{U_\infty} \, \mathrm{d}y}{\delta_2} \right) \frac{1}{U_\infty} \frac{\mathrm{d}U_\infty}{\mathrm{d}x} = \frac{\tau_\mathrm{w}}{\rho U_\infty^2}$$
(14)

From Eq. (5), we have $\frac{5}{4}HU_{\infty} = BV_0$ Then, we get

$$H = 3.858 \frac{x}{Re} + \left(0.8 - \frac{0.6277}{Re^{0.305}}\right) B \tag{15}$$

$$U_{\infty} = V_0 / \left[4.82 \frac{x}{Re} + \left(1 - \frac{0.785}{Re^{0.305}} \right) \right]$$
 (16)

From region (I) to region (III), the four order polynomial are used to approach the temperature profile which satisfies the boundary conditions:

$$\begin{cases} T = T_{w}, & \frac{\partial^{2} T}{\partial y^{2}} = 0, & \text{as } y = 0 \\ T = T_{\infty}, & \frac{\partial T}{\partial y} = \frac{\partial^{2} T}{\partial y^{2}} = 0 & \text{as } y = \Delta \end{cases}$$
 (17–18)

The temperature profile is

$$\theta = \frac{T - T_{\infty}}{T_{w} - T_{\infty}} = 1 - 2\frac{y}{A} + 2\left(\frac{y}{A}\right)^{3} - \left(\frac{y}{A}\right)^{4} \tag{19}$$

The integral energy equation is

$$\frac{\int_0^x q \, \mathrm{d}x}{\rho U_\infty C_p} = \Delta_1 (T_\mathrm{w} - T_\infty) \tag{20}$$

where
$$\Delta_1 = \int_0^\infty \frac{u}{U_\infty} \frac{T - T_\infty}{T_w - T_\infty} \, \mathrm{d}y$$
 (21)

By the Fourier law, Eq. (19) yields the heat flux at wall

$$q = \frac{2k}{4}(T_{\rm w} - T_{\infty}) \tag{22}$$

By solving Eqs. 9 and (19)–(22) simultaneously, a fourorder equation for Δ in stagnation zone is obtained

$$\xi^3 - 0.6726\xi^4 = 1.267 \frac{\varepsilon}{Pr} \tag{23}$$

where $\xi = \Delta/\delta$, $\delta = 2.749 BRe^{-0.5}$ and $\varepsilon = \int_0^x q \, dx/qx$.

The solutions of Eq. (23) with the parameter $Pr\varepsilon^{-1}$ yields, within 5% errors, approximately

$$\xi = \begin{cases} 1.245\varepsilon^{0.47} P^{-0.47}; & 1.0 \leqslant Pr\varepsilon^{-1} < 3.0\\ 1.147\varepsilon^{0.38} P^{-0.38}; & 3.0 \leqslant Pr\varepsilon^{-1} < 10.0\\ 1.082\varepsilon^{1/3} P^{-1/3}; & 10.0 \leqslant Pr\varepsilon^{-1} \end{cases}$$
(24)

So, $Nu = 0.727 \xi^{-1} Re^{0.5}$

$$= \begin{cases} 0.584 \varepsilon^{-0.47} Pr^{0.47} Re^{0.5}; & 1.0 \leqslant Pr \varepsilon^{-1} < 3.0 \\ 0.634 \varepsilon^{-0.38} Pr^{0.38} Re^{0.5}; & 3.0 \leqslant Pr \varepsilon^{-1} < 10.0 \\ 0.672 \varepsilon^{-1/3} Pr^{1/3} Re^{0.5}; & 10.0 \leqslant Pr \varepsilon^{-1} \end{cases}$$
(25)

In region (II), Δ can be solved from Eqs. (10), (13), (19), (21) and (22) and obtained:

$$\Delta = 3.894B^{0.76}x^{0.24}\epsilon Re^{-0.5}Pr^{-1/3}$$
(26)

Consequently,

$$Nu = 0.5136\varepsilon^{-1} \left(\frac{B}{x}\right)^{0.24} Re^{0.5} Pr^{1/3}$$
 (27)

Similar to above calculations, Δ and Nu can be obtained for region (III) as follows:

$$\Delta = 2.924 \varepsilon^{1/3} B^{1/3} P r^{-1/3} R e^{-1/3}$$

$$\times \left[3.858 \frac{x}{Re} + \left(0.8 - \frac{0.6277}{Re^{0.305}} \right) B \right]^{2/3}$$
(28)

 $Nu = 0.684\varepsilon^{-1/3} Pr^{1/3} Re^{1/3}$

$$\times \left[3.858 \frac{x}{ReB} + \left(0.8 - \frac{0.6277}{Re^{0.305}} \right) \right]^{-2/3}$$
 (29)

From Eqs. (15) and (28), the lateral position x_t can be determined as

$$x_{\rm t} = \left(\frac{6.48\varepsilon}{Pr} - 0.207Re + 0.163Re^{0.695}\right)B\tag{30}$$

In region (IV), when lateral distance x is larger than x_t , both flow and thermal boundary layers are fully developed through the whole liquid film. In this region, the free surface temperature of liquid film varies with x. Another temperature profile is given by satisfying the boundary condition of heat flux at wall:

$$\theta = 1 - \alpha \frac{y}{H} + \frac{1}{3} \alpha \left(\frac{y}{H}\right)^3 \tag{31}$$

where $\alpha = Hq/[k(T_w - T_\infty)]$.

By similar calculations, we can get

$$Nu = \left[\frac{2\varepsilon}{PrRe} + \frac{272H}{525B}\right]^{-1} \tag{32}$$

where H can be obtained from Eq. (15).

If the heat flux at the wall is uniform, q = constant or $\varepsilon = 1$, the heat transfer coefficients can be simplified from region I to region IV.

3. Comparison of analytical and experimental results

Theoretical expressions of heat transfer coefficients for free-surface slot jets were obtained by undertaking the mathematical analyses shown above. Comparisons of heat transfer coefficients in both the stagnation zone and the wall-jet zone were made between the present study and the analytical and experimental results in published literature.

Heat transfer coefficients at the stagnation line for slot jet impingement are shown in Fig. 2, in which both equations from this study and expressions from other researches are plotted. McMurray et al. [19], Miyasaka et al. [20], Zumbrunnen et al. [21] and Vader et al. [22] presented either analytical or empirical correlations of heat transfer for free-surface slot jet impingement. In their experiments water was mostly used as the working fluid together with nozzles of different slot widths. Good agreements between present theoretical results and the correlations from other studies were observed even for the high Reynolds numbers (Re > 10000). A deviation below 25% between the present equations and other empirical correlations exists except for those by Zumbrunnen with $B = 10.2 \,\mathrm{mm}$ due to different turbulence level of jet flow. Another plotted line based on the theoretical equation by Kendoush [15], also shown in the figure, shows somewhat a higher Nusselt numbers than

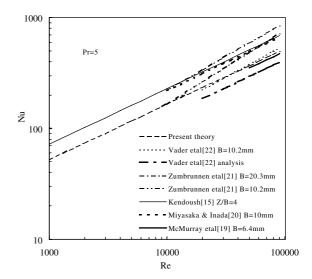


Fig. 2. Comparisons of stagnation heat transfer for free-surface slot jet impingement.

that of present theoretical results within a deviation of 27%. The relative deviations can be explained by the influence of the turbulent intensity of the free jet in those experiments. The level of turbulent flow at the exit of the nozzle is distinctly different for varied Prandtl numbers, Reynolds numbers and geometry of nozzles.

Figs. 3 and 4 exhibit comparisons of local heat transfer coefficients along heated surfaces between present results and the experimental data by Vader et al. [22], Zumbrunnen et al. [21]. The experimental results of Vader et al. [22] and Zumbrunnen et al. [21] with water as the working fluid are shown in Figs. 3 and 4, in which

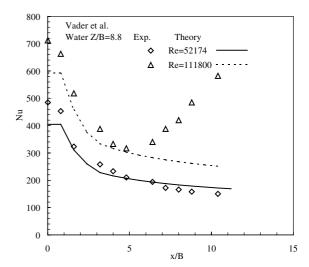


Fig. 3. Comparison of theoretical results with experimental results by Vader et al. [22] with water.

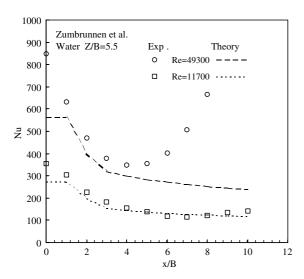


Fig. 4. Comparison of theoretical results with experimental results by Zumbrunnen et al. [21] with water.

generally good agreements between experimental results and the present correlations are observed before the transition from laminar to turbulent flow. As illustrated in the figures, the humps in heat transfer distribution curves are clearly observable from the experimental results, and are due to the transition at flow boundary layer from laminar flow to turbulence. The local heat transfer may be well regulated by the present theory, focused on the laminar flow of boundary layer, even for initially turbulent impinging jet with high Reynolds numbers (Re > 10000).

4. Conclusion

A theoretical analysis has been conducted to study heat transfer characteristics of free-surface liquid slot jet impingement. Equations for local heat transfer coefficients are obtained along heated surfaces with slot jets impingement based on the theory of laminar boundary layer. Also, the formula is compared with experimental results for both different working fluids and different nozzle sizes. Present theory shows good agreements with experimental results of for slot jet impingement. Further research should be developed to correlate heat and mass transfers of slot jet impingement with high turbulence.

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